



AEROSPACE MATERIAL SPECIFICATION

AMS4330™**REV. D**

Issued	2001-11
Reaffirmed	2012-04
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Superseding AMS4330C

Aluminum Alloy, Plate (2297-T87)
2.8Cu - 1.5Li - 0.30Mn - 0.12Zr
Solution Heat Treated, Stretched, and Artificially Aged

(Composition similar to UNS A92297)

RATIONALE

AMS4330D restores this document to active status, prohibits unauthorized exceptions Form (1.1), Exceptions (3.6), Reports (4.4.1), and Identification (5.1.1), updates fracture toughness to a standard test procedure (3.3.2.4), and results from a Five-Year Review and update of this specification.

1. SCOPE

1.1 Form

This specification covers an aluminum alloy in the form of plate 1.500 to 6.000 inches (38.1 to 152.40 mm) thick (see 8.5).

1.2 Application

These products have been typically used in aerospace applications where low density is needed in combination with moderate strength, high fatigue resistance, good stress corrosion properties, and improved stiffness, but usage is not limited to such applications.

2. APPLICABLE DOCUMENTS

The issue of the following documents in effect on the date of the purchase order form a part of this specification to the extent specified herein. The supplier may work to a subsequent revision of a document unless a specific document issue is specified. When the referenced document has been cancelled and no superseding document has been specified, the last published issue of that document shall apply.

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<https://www.sae.org/standards/content/AMS4330D/>

2.1 SAE Publications

Available from SAE International, 400 Commonwealth Drive, Warrendale, PA 15096-0001, Tel: 877-606-7323 (inside USA and Canada) or +1 724-776-4970 (outside USA), www.sae.org.

AMS2355 Quality Assurance, Sampling and Testing, Aluminum Alloys and Magnesium Alloy Wrought Products (Except Forging Stock), and Rolled, Forged, or Flash Welded Rings

AMS2772 Heat Treatment of Aluminum Alloy Raw Materials

ARP1917 Clarification of Terms Used in Aerospace Metals Specifications

2.2 ASTM Publications

Available from ASTM International, 100 Barr Harbor Drive, P.O. Box C700, West Conshohocken, PA 19428-2959, Tel: 610-832-9585, www.astm.org.

ASTM B594 Ultrasonic Inspection of Aluminum-Alloy Wrought Products for Aerospace Applications

ASTM B645 Plane-Strain Fracture Toughness Testing of Aluminum Alloys

ASTM B660 Packaging/Packing of Aluminum and Magnesium Products

ASTM B666/B666M Identification Marking of Aluminum and Magnesium Products

ASTM E399 Plane-Strain Fracture Toughness of Metallic Materials

ASTM G34 Exfoliation Corrosion Susceptibility in 2XXX and 7XXX Series Aluminum Alloys (EXCO Test)

ASTM G47 Determining Susceptibility to Stress-Corrosion Cracking of 2XXX and 7XXX Aluminum Alloy Products

2.3 ANSI Accredited Publications

Copies of these documents are available online at <http://webstore.ansi.org/>.

ANSI H35.1/H35.1M Standard Alloy and Temper Designation System for Aluminum

ANSI H35.2 Dimensional Tolerances for Aluminum Mill Products

ANSI H35.2M Dimensional Tolerances for Aluminum Mill Products (Metric)

3. TECHNICAL REQUIREMENTS

3.1 Composition

Shall conform to the percentages by weight shown in Table 1, determined in accordance with AMS2355.

Table 1 - Composition

Element	Min	Max
Silicon	--	0.10
Iron	--	0.10
Copper	2.5	3.1
Manganese	0.10	0.50
Magnesium	--	0.25
Zinc	--	0.05
Titanium	--	0.12
Lithium	1.1	1.7
Zirconium	0.08	0.15
Other Elements, each	--	0.05
Other Elements, total	--	0.15

3.2 Condition

Solution heat treated at 980 to 1000 °F (536 to 538 °C) and rapidly cooled in a suitable quenching medium; stretched to produce a permanent set of at least 2.75% but not more than 6.75%; artificially aged to T87 temper (refer to ANSI H35.1/H35.1M) at 320 °F (160 °C) for 20 to 48 hours. Aging time shall be dependent upon section thickness composition, and prior processing. Heat treatment shall be performed in accordance with AMS2772.

3.2.1 Plate shall receive no further straightening operations after stretching.

3.3 Properties

Plate shall conform to the following requirements, determined on the mill produced size in accordance with AMS2355 and as specified herein.

3.3.1 Tensile Properties

Shall be as shown in Table 2.

Table 2A - Minimum tensile properties, inch/pound units

Temper	Thickness, Inclusive (Inches)	Grain Direction	Ultimate Tensile Strength (ksi)	Yield Strength 0.2% Offset (ksi)	% Elongation in 2 Inches or 4D (%)
T87	1.500 to 2.000	L	64.0	58.0	10
		LT	66.0	60.0	8
		ST	65.0	57.0	2
	2.001 to 2.500	L	63.0	57.0	9
		LT	65.0	59.0	7
		ST	64.0	56.0	2
	2.501 to 3.000	L	62.0	57.0	9
		LT	64.0	58.0	7
		ST	62.0	55.0	2
	3.001 to 4.000	L	62.0	57.0	5
		LT	62.0	57.0	4
		ST	59.0	54.0	1.5
	4.001 to 5.000	L	61.0	56.0	5
		LT	61.0	56.0	4
		ST	58.0	52.0	1.5
	5.001 to 6.000	L	60.0	55.0	5
		LT	60.0	55.0	4
		ST	57.0	52.0	1.5

Table 2B - Minimum tensile properties, SI units

Temper	Thickness, Inclusive (Millimeters)	Grain Direction	Ultimate Tensile Strength (MPa)	Yield Strength 0.2% Offset (MPa)	% Elongation in 50.8 mm or 4D (%)	% Elongation in 5D or 5.65 √A (%)
	38.1 to 50.8	L	441	400	10	9
		LT	455	414	8	7
		ST	448	393	2	2
	50.9 to 63.5	L	434	393	9	8
		LT	448	407	7	6
		ST	441	386	2	2
	63.4 to 76.10	L	427	393	9	8
		LT	441	400	7	6
		ST	427	379	2	2
	76.20 to 101.60	L	427	393	5	4
		LT	427	393	4	3
		ST	407	372	1.5	1.5
	101.62 to 127.00	L	421	386	5	4
		LT	421	386	4	3
		ST	400	358	1.5	1.5
	127.02 to 152.40	L	414	379	5	4
		LT	414	379	4	3
		ST	393	358	1.5	1.5

3.3.2 Corrosion-Resistance

3.3.2.1 Stress-Corrosion Resistance

Direct tension specimens machined and tested in accordance with ASTM G47 shall show no evidence of stress corrosion failure when stressed in the short-transverse direction at 30.0 ksi (207 MPa) and exposed for 30 days.

3.3.2.2 Exfoliation Corrosion Resistance

When required by the purchaser, specimens from preproduction plates shall show exfoliation corrosion resistance equal to or better than EB when tested at the T/10 plane in accordance with ASTM G34.

3.3.2.3 Plate not meeting the requirements of 3.3.2.1 or 3.3.2.2 may be given additional precipitation heat treatment. After such treatment, if all specified properties are met, the plate is acceptable.

3.3.2.4 Fracture Toughness

Plane strain fracture toughness shall be determined in accordance with ASTM E399 and ASTM B645. A valid K_{Ic} meeting the requirements of ASTM E399, or a K_Q "usable for lot release" in accordance with ASTM B645 shall meet the requirements shown in Table 3. For L-T and T-L test directions, use specimens with 1.5 inch (38.1 mm) minimum thickness. For the S-L test direction, specimen thickness should be maximized to the nearest 0.25 inch (6.35 mm). L-T, T-L, and S-L specimens shall be centered at T/2 for plate 3.000 inches (76.2 mm) and under in nominal thickness and centered at T/4 for plate over 3.000 inches (76.2 mm) in nominal thickness. If a valid K_{Ic} or K_Q "usable for lot release" cannot be obtained due to insufficient specimen ligament size ($W - a$) or crack length, then the fracture toughness (K_Q) of the specimen will be acceptable for lot release if the following conditions are met:

- 3.3.2.4.1 The B dimension of the specimen tested shall be the maximum possible up to 2.5 inches (63.5 mm) for the given plate thickness.
- 3.3.2.4.2 The specimen centerline location shall be maintained at the specified plate thickness location.
- 3.3.2.4.3 All fracture toughness validity checks except specimen thickness, crack length or P_{max}/P_Q shall meet the criteria of ASTM E399 or ASTM B645.
- 3.3.2.4.4 If the only invalidity with $B = 1.5$ inches (38.1 mm) is P_{max}/P_Q , test result is acceptable for lot release.

Table 3A - Minimum fracture parameters, inch/pound units

Temper	Thickness, Inclusive (Inches)	Specimen Orientation	Required Minimum Plane- Strain Fracture Toughness K_{Ic} (ksi-in ^{1/2})
T87	1.500 to 3.000	L-T	32
		T-L	27
		S-L	20
	3.001 to 4.000	L-T	31
		T-L	27
		S-L	20
	4.001 to 5.000	L-T	30
		T-L	26
		S-L	18
	5.001 to 6.000	L-T	29
		T-L	25
		S-L	18

Table 3B - Minimum fracture parameters, SI units

Temper	Thickness, Inclusive (Millimeters)	Specimen Orientation	Required Minimum Plane- Strain Fracture Toughness K_{Ic} (MPa-m ^{1/2})
T87	38.10 to 76.10	L-T	35
		T-L	30
		S-L	22
	76.20 to 101.60	L-T	34
		T-L	30
		S-L	22
	101.62 to 127.00	L-T	33
		T-L	29
		S-L	20
	127.02 to 152.40	L-T	32
		T-L	28
		S-L	20

3.4 Quality

Plate, as received by purchaser, shall be uniform in quality and condition, sound, and free from foreign materials and from imperfections detrimental to usage of the plate. Light scratches and discoloration or streaking shall not be reason for rejection.

3.4.1 Plate shall be ultrasonically inspected in accordance with ASTM B594 and shall meet the requirements of Class A.

3.5 Tolerances

Shall conform to all applicable requirements of ANSI H35.2 or ANSI H35.2M.

3.6 Exceptions

Any exceptions shall be authorized by purchaser and reported as in 4.4.1.

4. QUALITY ASSURANCE PROVISIONS

4.1 Responsibility for Inspection

The producer of plate shall supply all samples for producer's tests and shall be responsible for the performance of all required tests. Purchaser reserves the right to sample and to perform any confirmatory testing deemed necessary to ensure that the plate conforms to specified requirements.

4.2 Classification of Tests

4.2.1 Acceptance Tests

Composition (3.1) is an acceptance test and shall be performed on each ingot or each group of ingots poured simultaneously from the same source of molten metal. Tensile properties (3.3.1), fracture toughness (3.3.2.4), and ultrasonic inspection (3.4.1) are acceptance tests and shall be performed on each parent plate.

4.2.2 Periodic Tests

Stress-corrosion resistance (3.3.2.1) is a periodic test and shall be performed at a frequency selected by the producer unless frequency of testing is specified by purchaser.