## AERONAUTICAL MATERIAL SPECIFICATION

Society of Automotive Engineers, Inc. 29 West 39th Street **New York City** 

AMS 4121A

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ALUMINUM ALLOY BARS, ROLLED 4.5Cu - 0.9Si - 0.8Mn - 0.5Mg (14S-T6)

- 1. ACKNOWLEDGMENT: A vendor shall mention this specification number and its revision letter in all quotations and when acknowledging purchase orders.
- FORM: Rods, bars, and shapes.
- APPLICATION: Primarily for parts requiring high strength and whose fabrication does not involve welding or forming.
- COMPOSITION:

3.9 Copper - 1.2 045 Silicon **0.40 - 1.2** Manganese 0.20 - 0.8Magnesium 1.0 max Iron 0.25 max Zino 0.15 Titanium max Chromium 0.10 mar 0.05 max Other Impurities, each 0.15 max Other Impurities, total remainder Aluminum

- Solution and precipitation heat treated. CONDITION:
- TECHNICAL REQUIREMENTS:
- Tensile Properties:

65,000 min Tensile Strength, psi Yield Strength at 0.2% Offset or at 0.0145 inch in 2 in. Extension Under Load (E=10,500,000), psi Elongation, % in 4D

55,000 min 8 min

- 6.1.1 Tensile properties of material under 0.170 in. in diameter or distance between parallel sides shall be as agreed upon by purchaser and vendor.
- 6.2 Hardness: Material should have hardness not lower than Brinell 125 using 500 kg load and 10 mm ball or 1000 kg load and 9/16 in. ball, or not lower than Brinell 130 using 1000 kg load and 10 mm ball, but shall not be rejected on the basis of hardness if the tensile property requirements are met.
- QUALITY: Material shall be uniform in quality and condition, clean, sound, smooth, and free from foreign materials and from internal and external defects detrimental to fabrication or to performance of parts.